Platform Machining Evaluation

Project Number #NP06009505 Contract Number #W31P4Q-05-D-R003

Final Report

Revision 3 – September 21, 2006

Air Methods Englewood, Co.

Submitted by **Doug Perillo**



National Center for Defense Manufacturing & Machining
Doug Perillo, Project Engineer
1600 Technology Way
Latrobe, PA 15650
(724) 539-5901 Phone
(724) 539-5132 fax
www.ncdmm.org

Report Documentation Page			Form Approved OMB No. 0704-0188		
Public reporting burden for the collection of information maintaining the data needed, and completing and review including suggestions for reducing this burden, to Wash VA 22202-4302. Respondents should be aware that not does not display a currently valid OMB control number	ving the collection of information. Send comme ington Headquarters Services, Directorate for I withstanding any other provision of law, no per	nts regarding this burden estin nformation Operations and Re	nate or any other aspect ports, 1215 Jefferson I	et of this collection of information, Davis Highway, Suite 1204, Arlington	
1. REPORT DATE 21 SEP 2006	2. REPORT TYPE Final		3. DATES COVERED 19-05-2006 to 21-09-2006		
4. TITLE AND SUBTITLE Platform Machining Evaluation Air Methods			5a. CONTRACT NUMBER W31P4Q-05-D-R003		
			5b. GRANT NUMBER		
			5c. PROGRAM I	ELEMENT NUMBER	
6. AUTHOR(S) Doug Perillo			5d. PROJECT NUMBER 06-0095-05		
		5e. TASK NUMBER			
			5f. WORK UNIT	NUMBER	
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) National Center for Defense Manufacturing & Machining,1600 Technology Way,Latrobe,PA,15650			8. PERFORMING ORGANIZATION REPORT NUMBER		
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)			10. SPONSOR/MONITOR'S ACRONYM(S)		
			11. SPONSOR/M NUMBER(S)	MONITOR'S REPORT	
12. DISTRIBUTION/AVAILABILITY STAT Approved for public release; d					
13. SUPPLEMENTARY NOTES					
Air Methods Corporation, Eng MEDEVAC (Medical Evacuat pieces designed to support litte ambulatory patient seats. The with the pending introduction and the loads on the platforms National Center for Defense M and develop a manufacturing of the weight of the platform alon	ion) variant of the UH-60 er-borne patients. These poriginal platforms were dof the HH-60M, the platfothemselves have increase lanufacturing and Machiporocess capable of produc	Black Hawk into latforms also ser esigned to meet o orms must meet s d. Air Methods (ning (NCDMM) n ing a complex sti	erior that are ve as attache civil crash re stricter Arm Corporation review a pro	e machined aluminum ment points for the equirements. However, y crash requirements, requested that the posed stiffener design	
15. SUBJECT TERMS National Center for Defense M Corporation; Kennametal Inc.	U	ning; UH-60Q M	EDEVAC; A	Air Methods	
16. SECURITY CLASSIFICATION OF:		17. LIMITATION	18. NUMBER	19a. NAME OF	

OF ABSTRACT

1

c. THIS PAGE

unclassified

a. REPORT

unclassified

b. ABSTRACT

unclassified

OF PAGES

13

RESPONSIBLE PERSON

Air Methods **Platform machining Evaluation**

Prepared by: Doug Perillo

Executive Summary

In 1996, the US Army began fielding the UH-60Q MEDEVAC (Medical

Evacuation) variant of the UH-60 Black Hawk. The MEDEVAC variant

fielding has continued in low-rate production since the HH-60L variant, but

has remained unchanged since its initial design. With the introduction of the

UH-60M into the Army's fleet, the MEDEVAC fleet must keep pace to meet

the requirements of Aviation Transformation. With the pending introduction

of the new HH-60M version, the Army is faced with the conflicting goals of

reducing system weight and unit cost while simultaneously meeting tougher

crash loading standards for the patient care systems.

The litter pans (platforms) used in the MEDEVAC interior are machined

aluminum pieces designed to support a litter-borne patient during a crash,

and to also serve as an attachment point for the ambulatory patient seats.

The original platforms were designed to meet civil crash requirements.

However, for the HH-60M, the platforms must meet stricter Army crash

requirements, and the loads on the platforms themselves have increased.

The contractor has proposed increasing the structural strength of the

platforms by increasing the web thickness on the underside, but this

improvement comes at a penalty of increased weight.

Reducing the weight of the Medical Interior increases the performance

margin on the aircraft. This translates into increased "payload" capacity,

better performance, better fuel efficiency, and longer aircraft life.

Project Number #NP06009505 Contract Number #W31P4Q-05-D-R003 Page 2 of 13

Therefore, Air Methods has requested that the National Center for Defense Manufacturing and Machining (NCDMM) review proposed stiffener designs and develop a manufacturing process capable of producing a complex stiffener geometry that will reduce the weight of the platform along with increasing its "payload" capacity.

Project Details

Air Methods along with the NCDMM engineers reviewed several improved strength stiffener designs. Air Methods decided on a design that would increase the bottom face contact surface of the platform. It is believed that this increased platform surface will satisfy the demand for the increased load on the UH-60Q Platform. The new design allowed for a 50% increase on the contact surface. The old design, consisting of a straight wall, will be replaced with an under cut wall with both the bottom and the top of the wall having radii to strengthen and blend to the wider surface, see Figure #1.

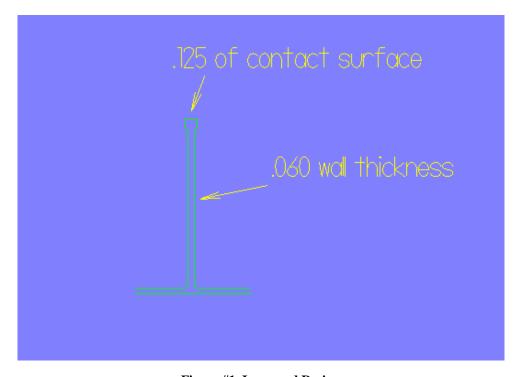


Figure #1, Improved Design

Based on the size of the UH-60Q platform, a team decision was made to perform a "Proof-of-Concept" on a scaled down, single section of the platform. The section would be 7" wide x 12" long x 1.75" thick. The machined area would include two pockets 1.690" deep. The wall and floor thickness would be maintained at .060", see Figure #2.

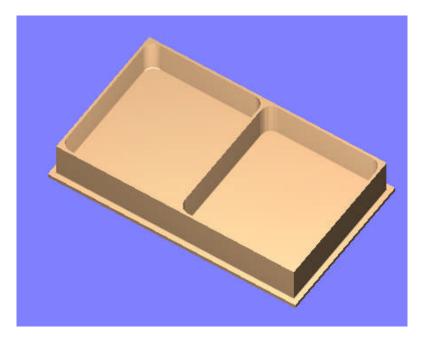
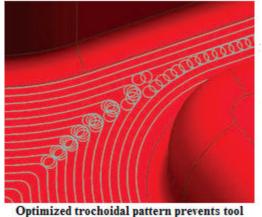


Figure #2, Part Representation

The NCDMM engineers modeled the part using Mastercam software, while developing the tool paths. High Speed Machining (HSM) techniques were used along with thin wall machining techniques. When using HSM techniques, lighter depths of cut (DOC) and faster feeds and speeds are used. HSM requires tool paths that maintain a constant chip load, see Figure #3. Smaller tools are used with high surface feet per minute (SFM). The use of smaller tools helps to reduce the residual stress (generated by the cutting action) from entering the part causing a warp condition.



Optimized trochoidal pattern prevents tool from exceeding intended metal removal rate.

Figure 3, HSM tool path.

Thin wall machining is a technique that is used on walls less than .100" thick. When machining the part, larger roughing stock values are left all around the part. On this particular part it was decided to leave .100" on all surfaces. This extra stock will then be removed during the finish cut. The tool used to thin wall machine is also modified with shank clearance. This clearance is required so that the tool will not rub as it steps down the finished wall, see Figure #4.

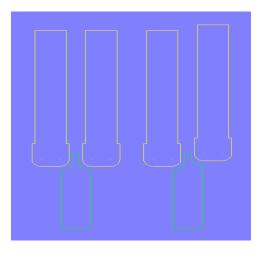


Figure #4, Visual of Thin Wall Machining

The additional stock strengthens the wall at the cutting point, which reduces the amount of push resulting from thin walls.

Using all of these techniques, the NCDMM technicians generated tool paths to rough out the complete part leaving .100" of material on the walls for the finish cut, see Figure #5.

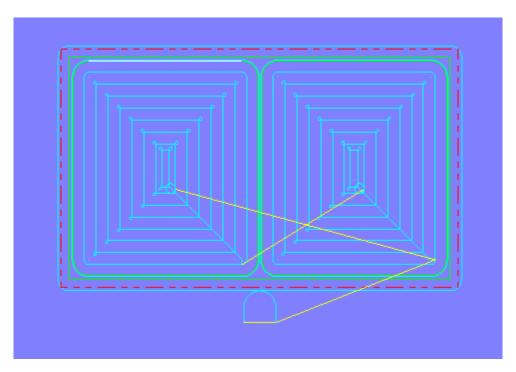


Figure #5, Roughing Tool Path

The selected roughing tool was a .500" diameter, three (3) flute, 37-degree helix end mill. This end mill is a standard Kennametal product and is designed for cutting aluminum. Clearance was ground on the shank to avoid rubbing during deep cutting, see Figure #6 and Figure #7.





Figure #6, Kennametal HPF37A

Figure #7, Shank Relief

The Kennametal HPF37A is a solid carbide end mill grade KC651M. The coating on the end mill is a TiB2, which is an extreamly hard coating that provides very good ware characteristic at high cutting speeds. It reduces edge built up and can help reduce burring. The roughing cycle was preformed at 1200 SFM, which equates to 9168 revolutions per minute (RPM). The axial depth of cut (ADOC) chosen equaled .200" and the radial width of cut (RWOC) chosen equaled 75% of the tool diameter. The feed was preformed at .004 inches per tooth (IPT), which results in 110.0 inches per minute (IPT)

The tool holder chosen for both roughing and finishing was a Kennametal Powergrip milling chuck, see Figure #8.



Figure #8, Powergrip Milling Chuck

The roughing cycle resulted in a 25minute run time. Each pocket along with the out side periphery were roughed complete to the step depth of .200" before moving to next step depth.

Finishing paths were developed in Mastercam software using a custom ground, HPF37A end mill. The tool was ground with a shank relief and the required .032" radius both on top and bottom, see Figure #9. This radius will achieve the required geometry on the top as well as the bottom of the recessed platform wall.



Figure #9, Custom Tool Geometry

The first step in the finishing process was to machine the walls to size. This was completed in .100" axial steps. The axial step depth is determined by the grind on the form tool. It should be noted that this could have been greater, which would result in less run time by reducing the number of passes. The finish cycle was preformed at 1000 SFM, which equates to 7640 RPM. The feed was preformed at .0025" IPT or 57" IPM. The tool path was similar to the roughing where each pocket and the outside periphery were machined complete to a predetermined step depth before moving to next step depth, see Figure #10.

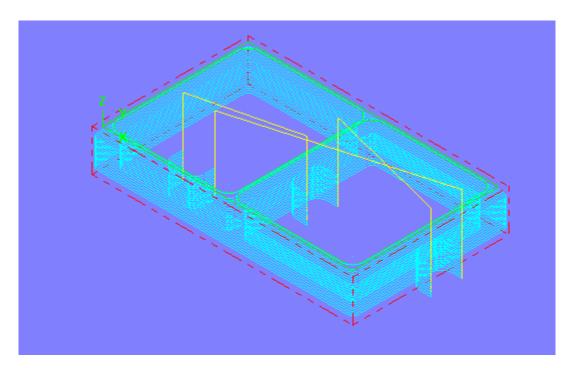


Figure #10, Finish Tool Path

The .100" axial step was repeated to the depth of 1.590", allowing for the .100" stock on the floor.

The next step in the finishing process required machining the .100" stock from the floor. Using another thin wall technique, this was completed in each pocket resulting in the required .060" floor thickness. The technique requires machining from the center of the pocket outward towards the pocket walls. This method assures that all the machining is taking place at the strongest area on the floor; see Figure #11 and Figure #12.

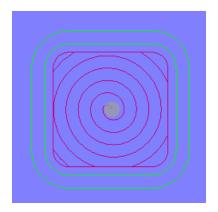


Figure #11, Example of Thin Floor Machining

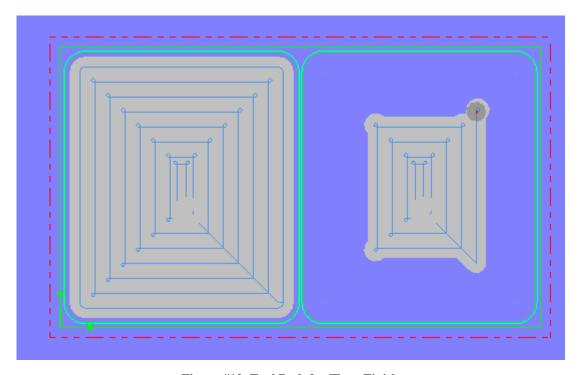


Figure #12, Tool Path for Floor Finish

The floor was then finished to a depth of 1.690". The floor path blended the floor to the walls with the required .032" radius.

Conclusion

Using state-of-the-market technology and methods described above, the scaled down version of the platform was machined resulting in a total run time of 55 minutes. The NCDMM feels that continuing the testing and tool path refinement, the total run can be further reduced. It is believed that this increased platform surface will satisfy the Army's demand for the increased "payload" on the UH-60Q MEDEVAC Platform. This new design could potentially reduce the weight of the platform by 30%, in some areas, and also allow for a 50% increase on the contact surface over the old, straight wall design. The "Proof-of-Concept" produced by the NCDMM, from this new design, met all the required geometric part tolerances; see Figure #13 thru Figure #15.

The NCDMM also recommends that several improved platform styles be compared through Finite Element Analysis (FEA). The NCDMM would be able to review any platform styles developed by Air Methods and also provide assistance in that development. The NCDMM would also be able to provide recommendations on any machinability issues that may arise during the machining of the new platform styles.

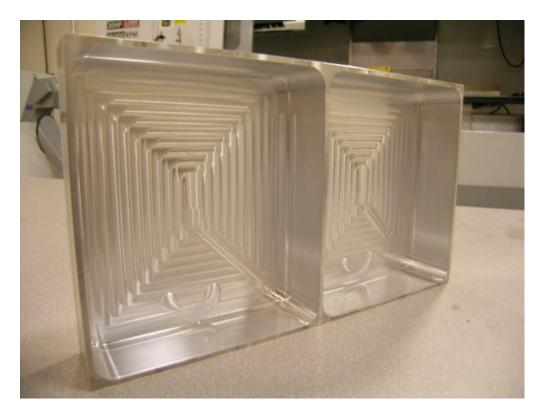


Figure #13, Finished Part

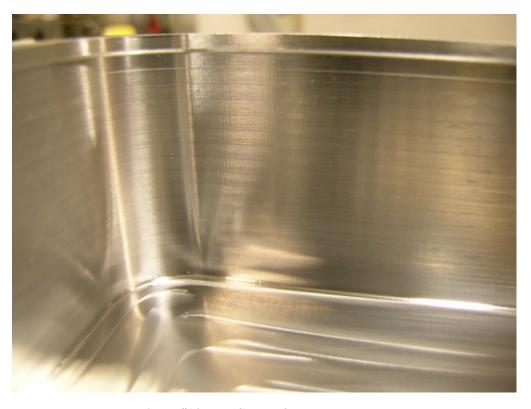


Figure #14, Part Corner & Recessed Wall

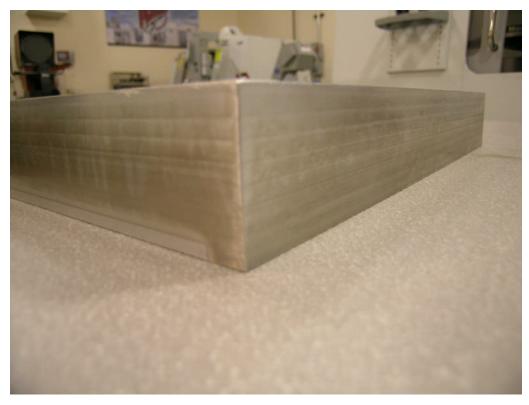


Figure #15, Outside Periphery of Part

This technical report made possible by the
United States Army
Army Material Command (AMC)
Research, Development and Engineering Command (RDECOM)
Aviation and Missile Research, Development and Engineering Center (AMRDEC)
Engineering Directorate (ED)
Manufacturing Science and Technology Division (MST)
Redstone Arsenal, Alabama
Scott A. Hofacker, PE – Program Manager 256.842.7992

scott.hofacker@us.army.mil
or Mike Cummings (Tiburon) 256.313.6496
mike.a.cummings@us.army.mil